AMENDMENTS TO THE SPECIFICATION:

Please amend the title as follows:

METHOD FOR AN AIRCRAFT NAVIGATIONAL AID AND CORRESPONDING DEVICE

On page 1 after the title, please insert the following heading and paragraph:

CROSS - REFERENCE TO RELATED APPLICATIONS

The present Application is based on International Application No. PCT/FR2003/002001, filed on June 27, 2003, which in turn corresponds to FR 02/08470 filed on July 5, 2002, and priority is hereby claimed under 35 USC §119 based on these applications. Each of these applications are hereby incorporated by reference in their entirety into the present application.

On page 1 before the first paragraph, please insert the following heading:

FIELD OF THE INVENTION

On page 1 after the fourth paragraph, please insert the following heading:

DESCRIPTION OF THE PRIOR ART

On page 3 after the second paragraph, please insert the following heading:

SUMMARY OF THE INVENTION

On page 4 after the third paragraph, please insert the following heading:

BRIEF DESCRIPTION OF THE DRAWINGS

On page 5 after the sixth paragraph, please insert the following heading:

DETAILED DESCRIPTION OF THE INVENTION

On page 13 after the third paragraph, please insert the following:

It will be readily seen by one of ordinary skill in the art that the present invention fulfills all of the objects set forth above. After reading the foregoing specification, one of ordinary skill will be able to affect various changes, substitutions of equivalents and various other aspects of the invention as broadly disclosed herein. It is therefore intended that the protection granted hereon be limited only by the definition contained in the appended claims and equivalents thereof.

Abstract:

Please amend the current Abstract as follows:

3

Docket No.: 4590-367

ABSTRACT

AIRCRAFT NAVIGATION AID METHOD AND CORRESPONDING DEVICE

The invention relates to an aircraft navigation aid method. It comprises the following steps consisting in: of [[a)]]computing a feeler line according to the wind, in other words the ground path that the aircraft would follow if a turn at the maximum rate applicable to the current flight phase of the aircraft were to begin at that instant, and [[b)]]displaying on a navigation screen the feeler line and a ground path to be captured, in order to determine how to place the aircraft in a turn in order to optimize the capture of the path to be captured.

No figure

4